

Incompressible Flow around a body



Useful CAE tool for aerodynamics studies

- Aerodynamics design for cars.
- Air flow analysis for heat problems inside PC.
- Environmental studies of the wind around buildings.

We are interested in the interaction
between flow and bodies (materials)

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Discretization for u, v and p at the x-wall



Arakawa-A:

$$u_{i,j} = 0 \quad v_{i,j} = 0 \quad \frac{p_{i,j+1} - p_{i,j-1}}{2\Delta y} = 0$$

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Arakawa-B:

$$u_{i,j} = 0 \quad v_{i,j} = 0 \quad \frac{p_{i,j} - p_{i,j-1}}{\Delta y} = 0$$

$$\frac{p_{i-1,j} - p_{i-1,j-1}}{\Delta y} = 0$$

2

Arakawa-C:

$$u_{i,j} = -u_{i,j-1} \quad v_{i,j} = 0 \quad \frac{p_{i,j} - p_{i,j-1}}{\Delta y} = 0$$

$$u_{i+1,j} = -u_{i+1,j-1}$$

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Boundary Condition at the body surface



Non-slip velocity condition:

$$u = 0 \quad \frac{\partial p}{\partial y} = \frac{1}{Re} \frac{\partial^2 v}{\partial y^2} \approx 0$$

High Reynolds number

$$v = 0 \quad \frac{\partial p}{\partial x} = \frac{1}{Re} \frac{\partial^2 u}{\partial x^2} \approx 0$$

High Reynolds number

SMAC Algorithm :

$$\frac{\partial p}{\partial y} = 0 \quad (x\text{-Wall Boundary})$$

$$\frac{\partial p}{\partial x} = 0 \quad (y\text{-Wall Boundary})$$

Neumann Boundary condition

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Discretization for u, v and p at the y-wall



Arakawa-A:

$$u_{i,j} = 0 \quad v_{i,j} = 0 \quad \frac{p_{i+1,j} - p_{i-1,j}}{2\Delta x} = 0$$

$$p_{i-1,j} \quad p_{i,j} \quad p_{i+1,j}$$

Arakawa-B:

$$u_{i,j} = 0 \quad v_{i,j} = 0 \quad \frac{p_{i,j} - p_{i,j-1}}{\Delta x} = 0$$

$$\frac{p_{i-1,j} - p_{i-1,j-1}}{\Delta x} = 0$$

$$p_{i-1,j-1} \quad u_{i,j} \quad v_{i,j} \quad p_{i,j-1} \quad p_{i,j}$$

Arakawa-C:

$$u_{i,j} = 0 \quad v_{i,j+1} = -v_{i-1,j+1} \quad \frac{p_{i,j} - p_{i-1,j}}{\Delta x} = 0$$

$$v_{i,j} = -v_{i-1,j}$$

$$v_{i-1,j+1} \quad u_{i,j} \quad v_{i,j+1} \quad p_{i-1,j} \quad p_{i,j} \quad v_{i-1,j} \quad v_{i,j}$$

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Boundary Treatment for the pressure



Indicator: **iw** (at the same point as the pressure)

iw = 1 inside the body.

iw = 0 outside the body.

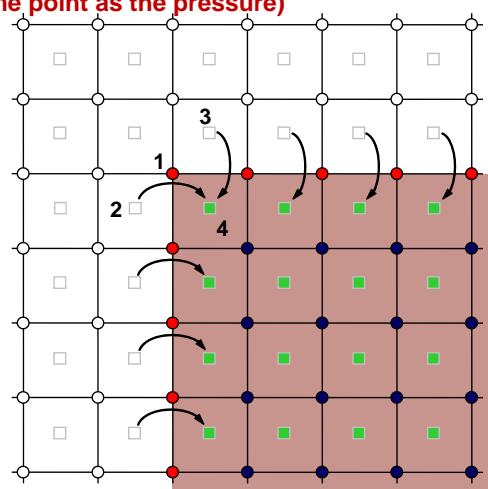
if $iw_{i+1,j} = 0$

when we solve $p_{i,j}$,

$$\frac{p_{i-1,j} - 2p_{i,j} + p_{i+1,j}}{\Delta x^2}$$

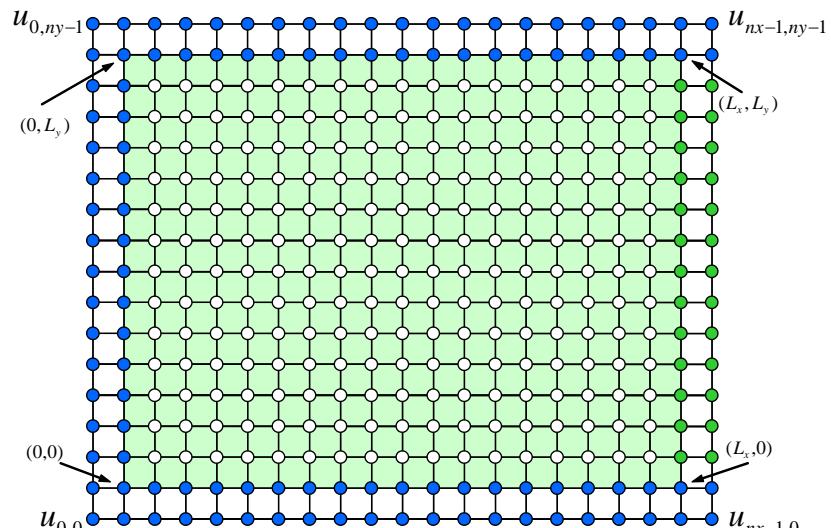
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$$\frac{p_{i-1,j} - 2p_{i,j} + p_{i,j}}{\Delta x^2}$$



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Boundary Condition for the Velocity



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Boundary Condition for Computational Domain

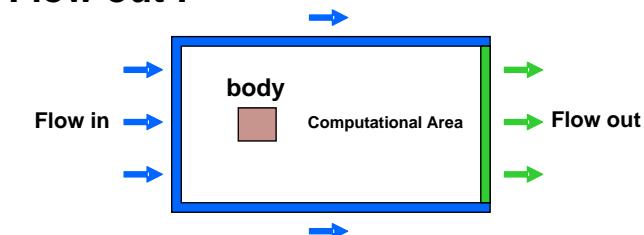


Computational domain : $0 \leq x \leq L_x \quad 0 \leq y \leq L_y$

Grid size : nx, ny

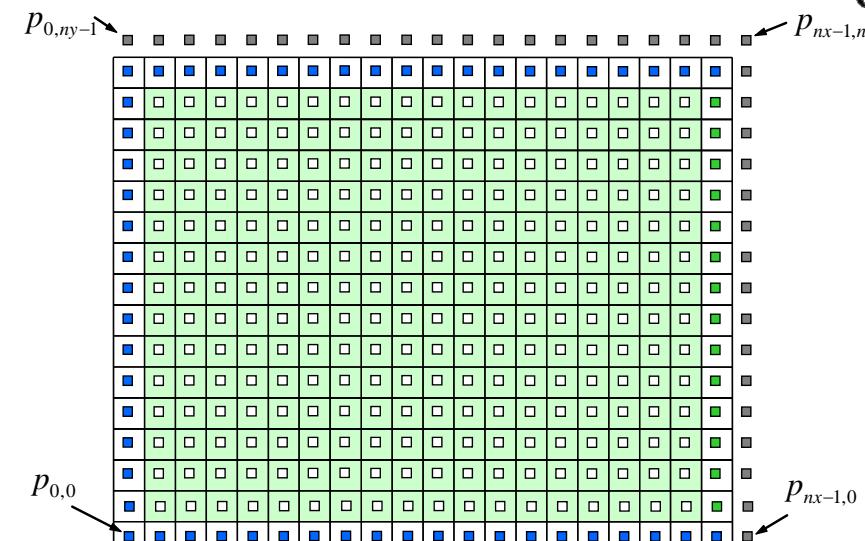
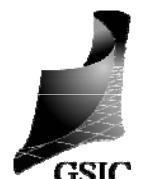
Grid spacing : $\Delta x = L_x / (nx - 1), \quad \Delta y = L_y / (ny - 1)$

Flow in / Flow out :



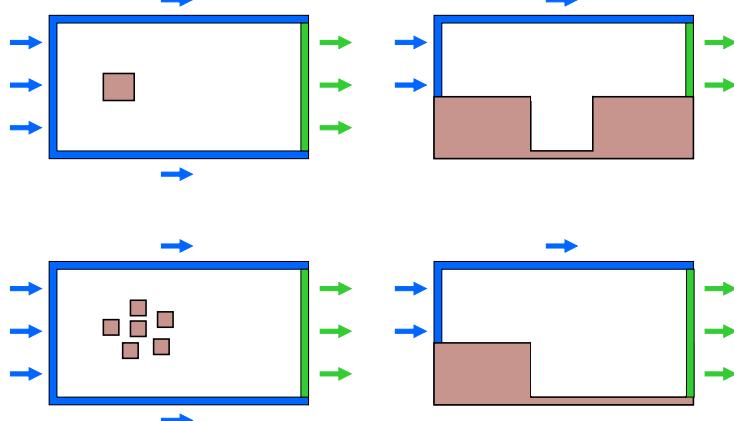
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Boundary Condition for the Pressure



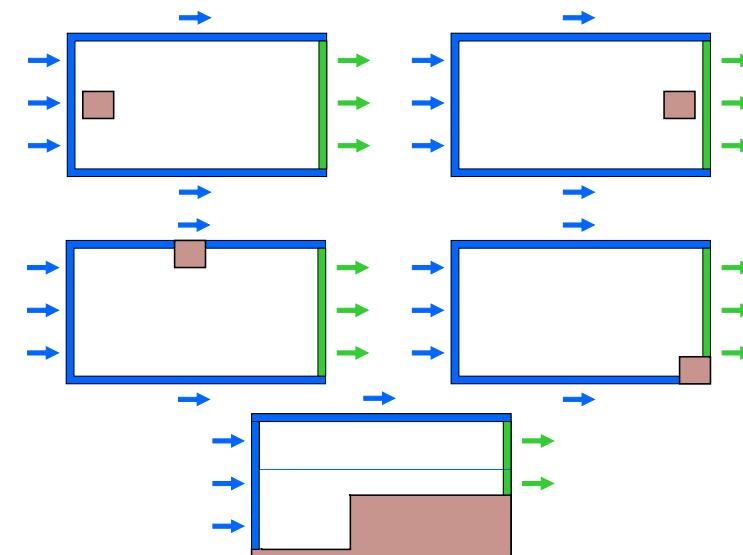
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Proper Layout of the bodies



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Bad Layout of the bodies



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Typical Computational Results



Source Code

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Submission of the Final Report



Modify the sample code and set the body of your favorite shape.

Show the numerical results (snap shots) and describe your consideration from the view points of both numerical calculation and fluid dynamics.

Deadline: 15th February, 2009

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